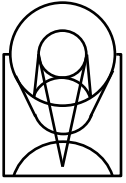




Science and Operations Center for JWST

J Pollizzi, A Krueger, G Green
ADASS 2009

Space Telescope Science Institute

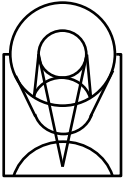


S&OC for the JWST

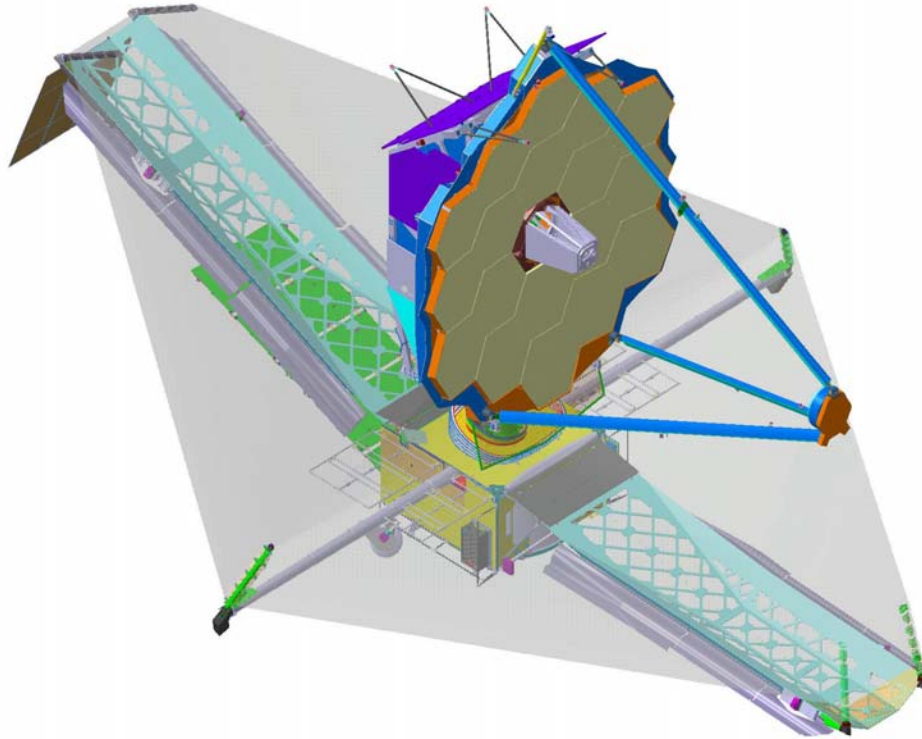


■ Agenda

- ◆ Brief background of the satellite and mission
- ◆ Lessons learned from operating the HST that we plan on applying to JWST
- ◆ Challenges we face in developing the S&OC



The James Webb Space Telescope



Effective Mirror Size: 6.5m; 18 segments

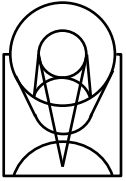
Sunshield Size: 14.6 m x 21.1 m

Payload Weight: 6500 kg (estimated)

Cold Side Operating Temp: -233C

Hot Side Operating Temp: +85C

Planned Consumables Lifetime: 10 years



JWST Teams and Responsibilities



Integrated Science Instrument Module (ISIM) – GSFC

- Structure – GSFC/ATK
- MIRI- JPL & ESA/European Consortium
- NIRSpec- ESA/Astrium
- NIRCам – U of Arizona/LMATC
- FGS/TFI – CSA/COM DEV

ISIM Radiators- NGST/Ball

Optical Telescope Element (OTE) – NGST/Ball

Backplane Structure – NGST/ATK

ISIM Electronics Compartment (IEC)- GSFC

Deployment Tower - NGST

Sunshield – NGST

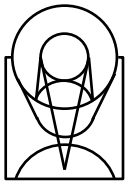
Membrane – NGST/SRS

Spacecraft - NGST

MIRI Cryocooler – JPL/NGST
Instrument C&DH - GSFC

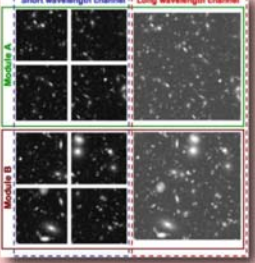
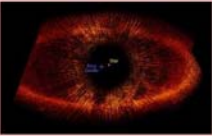
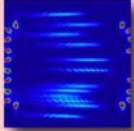
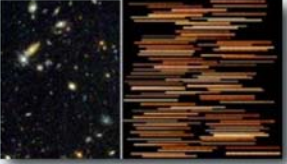
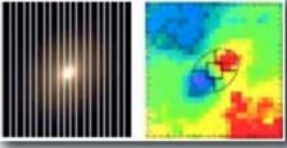

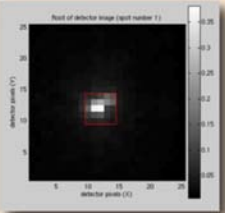


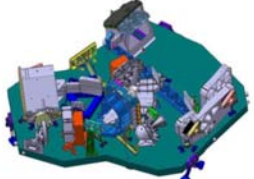
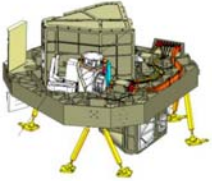
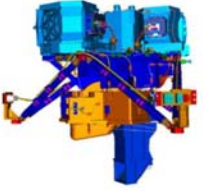


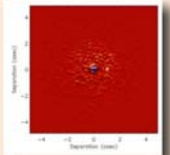

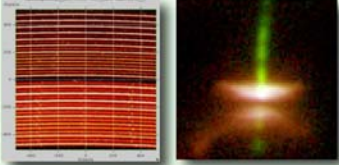
Launch Vehicle & Adapter- ESA/Arianespace

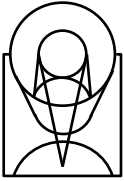
Overall Observatory - NGST
ADASS-2009



JWST Science Instruments

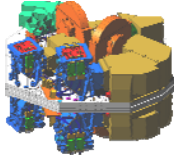

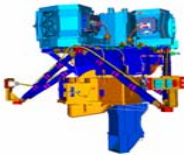
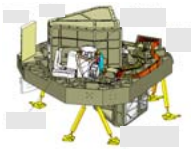


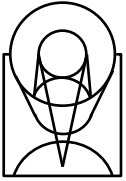
<p>Deep, wide field broadband-imaging</p>  <p>Short wavelength channel Long wavelength channel</p> <p>Module A Module B</p> <p>Coronagraphic Imaging</p> 	<p>Wavefront Sensing & Control (WFSC)</p> 	<p>Multi-Object, IR spectroscopy</p> 	<p>IFU spectroscopy</p>  <p>Long Slit spectroscopy</p> 
<p>Fine Guidance Sensor</p>  <p>Moving Target Support</p> 	<p>NIRCam</p>  <p>NIRSpec</p>  <p>FGS/TF</p>  <p>MIRI</p> 		<p>Mid-Infrared Imaging</p> 
<p>R=100 Narrowband Imaging</p>  <p>Coronagraphic Imaging R~100</p> 	<p>Mid-IR Coronagraphic Imaging</p> 	<p>IFU spectroscopy</p> 	



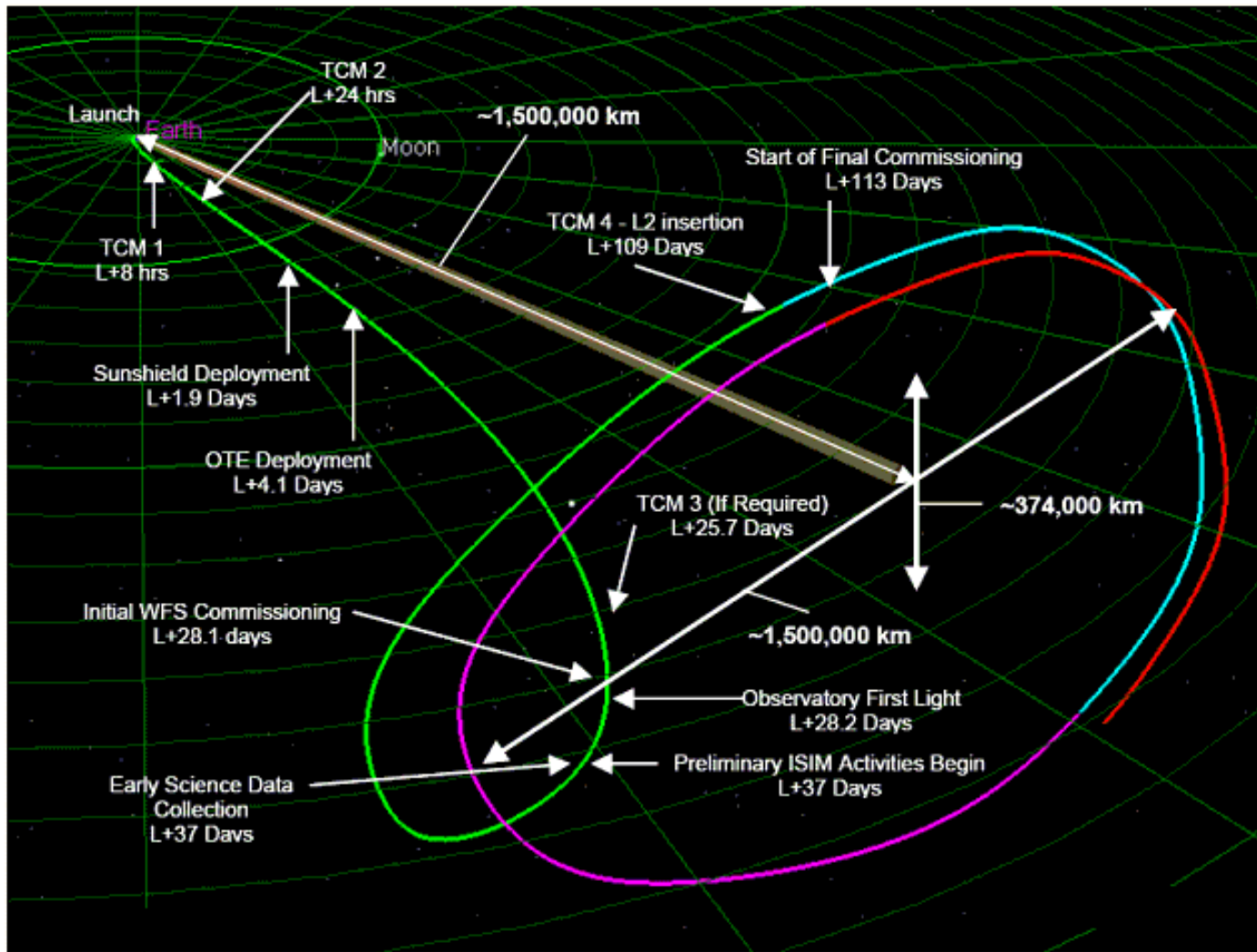
JWST Instruments

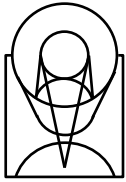


Instrument	Science Goal	Key Capability
NIRCam Univ. Az 	<i>Wide field, deep imaging</i> ⇒ 0.6 μm - 2.3 μm (SW) ⇒ 2.4 μm - 5.0 μm (LW)	Two 2.2' x 2.2' SW Two 2.2' x 2.2' LW
NIRSpec ESA 	<i>Multi-object spectroscopy</i> ⇒ 0.6 μm - 5.0 μm	9.7 Sq arcmin Ω 100 selectable targets R=100, 1000
MIRI ESA/JPL 	<i>Mid-infrared imaging</i> ⇒ 5 μm - 27 μm <i>Mid-infrared spectroscopy</i> ⇒ 4.9 μm - 28.8 μm	1.9' x 1.4' 3.7" x 3.7" - 7.1" x 7.7" R=3000 - 2250
FGS/TFI CSA 	<i>Fine Guidance Sensor</i> ⇒ 0.8 μm - 5 μm <i>Tunable Filter Imager</i> ⇒ 1.6 μm - 4.9 μm	Two 2.3' x 2.3' 2.2' x 2.2' R=100

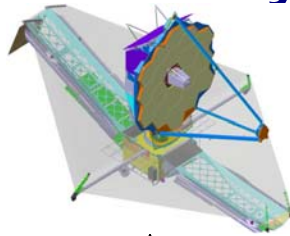


JWST Orbit Transfer





Key Operations Concepts



JWST Observatory

- L2 Orbit - thermal stability, stray light avoidance, stable communications
 - Orbit Determination & Maintenance
 - Momentum Management
- Event-Driven Operations - efficient and flexible operations
 - Parallel SI Calibrations - Darks, Sky Flats
- Wavefront Sensing & Control



Goldstone



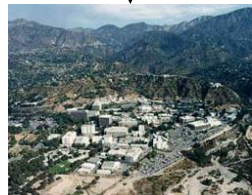
Madrid



Canberra

JPL Deep Space Network (DSN) Ground Stations

- CFDP File Transfer Protocol
- Ranging
 - Alternating hemispheres
- Clock Correlation
- Downlink rate
 - 458 gbits/day over 2x 4 hour passes
 - All science and engineering is recorded for downlink
 - Engr also downlinked real-time during pass



NASA Integrated Services Network (NISN) - MSFC - Huntsville AL
• Ground Network & Voice Communications



Goddard Space Flight Center (GSFC) - Greenbelt

- Flight Dynamics Facility (FDF)
- Flight Software Maintenance Facility



Space Telescope Science Institute (STScI) - Baltimore

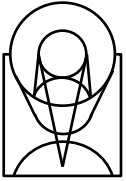
- Science Operations
 - Proposal Selection, Planning & Scheduling
 - Event-Driven Operations
 - Data Archive & Calibration
- Mission Operations
 - Real-Time Operations, Health & Safety Monitoring
 - Anomaly Resolution, Trending Analysis
- Normal Operations 8 x 5
 - Automated Contacts

JPL Deep Space Network (DSN) - Pasadena CA

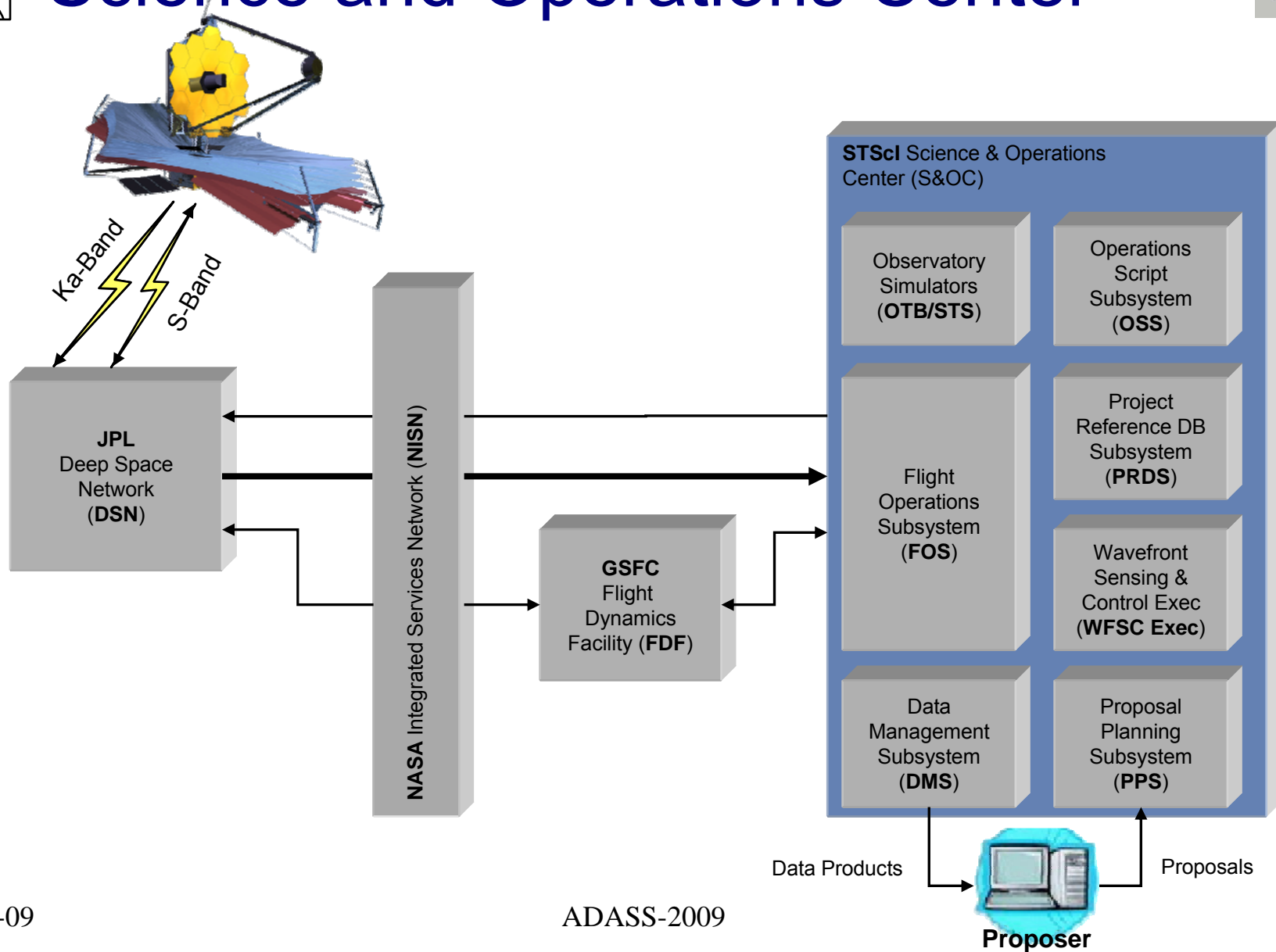
- Launch Support
- Commissioning Support - continuous coverage
- Normal Operations Support - 4 hour contact / day

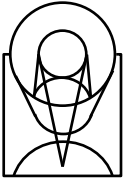
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Science and Operations Center

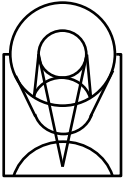




Lessons Learned from HST^(1/3)



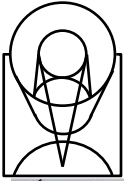
- Test as you fly
 - ◆ Use every opportunity and system used during I&T to work out flight operations concepts and approaches
 - ◆ Use testing as training events for eventual flight staff
- Take best of legacy systems – but not without true competitive analysis – and certainly not with any leftover baggage
 - ◆ E.g. keeping APT, SPIKE and some GSS –but not rest of planning
 - ◆ Likewise will use MAST, and our Storage Architecture (NSA), but not DADS and likely not OPUS
- Use an iterative design approach – but do a good requirements analysis upfront
 - ◆ The requirements gives both a list of what needs to be done as well as can drive the testing



Lessons Learned from HST^(2/3)



- Have a uniform mechanism for users to interact with Institute regardless of mission
 - ◆ Using APT as “the” proposal entry tool for both HST and JWST
 - ◆ Using MAST as the “front door” to the archive for all mission data held by STScI
 - ◆ Will use GMS as a consistent means for working with Grant submissions and reporting across missions



Example JWST APT Input



The screenshot displays the Astronomer's Proposal Tools (APT) Version 17.4 interface. The main window is titled "Observation 1 of JWST Proposal (v838-jwst.apt)".

Mosaic Configuration:

- Pattern Size: Rows: 2, Cols: 5
- Pattern Controls:
 - Row Overlap (%): 0.0
 - Col Overlap (%): 23.4
 - Skew X (Deg): 12.9
 - Skew Y (Deg): 0.0
 - Orient: 67.8

Table of Observations:

Observation	Number	Instrument	Instrument...	Coming So...	Object Type	Subarray	Filters	Total Time	Dither Pat...	Dither Point
Observati...	1	MIRI	MIRI Imagi...	Cyro,	BRIGHT	FULL	[F560W 2...	2001.12	Dither Pat...	[edu.stsci...

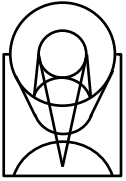
Multimission Archive at STScI (MAST) Panel:

- Target: 07:04:05.21 -03:50:48.5
- Radius: 2.5'
- Mission/Instrument: ALL - Search all missions

Aladin v5.0 Panel:

- ICRS coordinate: 07:04:05.21 -03:50:48.5
- Pixel size: 4570.0
- Field size: 15.0' x 14.29'
- Scale: 21.89"

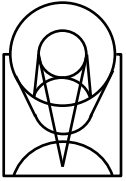
The main visualization shows a star field with a grid of observation patterns overlaid. The interface includes various toolbars for navigation and editing, and a taskbar at the bottom with various application icons.



Lessons Learned from HST^(3/3)



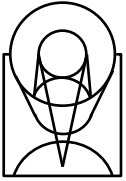
- Maximize efficiency by having onboard event driven operations
 - ◆ HST uses only absolute timing actions. Loses opportunities for more exposure time due to incurred overheads (i.e. waiting a fixed amount of time for a mechanism to move)
 - ◆ JWST uses event driven actions – i.e. can proceed with next step once a mechanism has reported motion completion
 - ◆ Event concept applies to visits as well. A succeeding visit can start as soon as its earlier visit successfully completes or fails (within some no-earlier-than constraints)
 - ◆ i.e. should a guide star acquisition fail, then that visit is canceled and the succeeding one can start. The science program can continue even in the wake of failed visits



Challenges^(1/5)



- Funding for Ground Segment components delayed
 - ◆ Defers startup of key aspects to ground system to different times
 - ◆ Impinges on “Test as you fly” since the software is not ready when the hardware is
 - ◆ And defers integration problems effectively till last major system comes online (will be years after the first system is ready)
 - Will have to rely more on intra-subsystem simulators/stubs to check out components

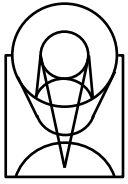


Challenges^(2/5)

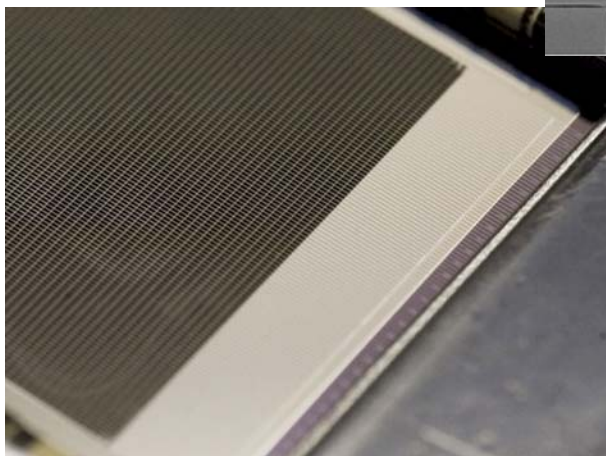
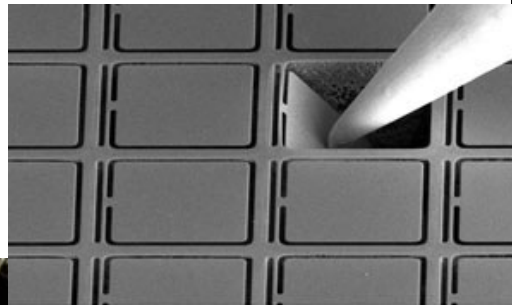


- Complexity of the Instruments and nature of the data
 - ◆ NIRSpec has a micro shutter array of 798 x 350 shutters (detector is 2 arrays of 2k x 2k); NIRCAM has 10 sensor arrays, each 2k x 2k
 - ◆ Data is downlinked as a cube holding successive read-outs as the sensors accumulate charge
 - ◆ Early indications imply a persistence effect on the detectors
 - ◆ Will require new approaches to doing the calibration pipelines

- Environment of the satellite
 - ◆ Solar wind applies a torque to the spacecraft
 - ◆ Momentum management requires an ongoing and active response
 - ◆ Factored into both the planning systems and onboard scripting system
 - Is complicated by the event driven nature of the visits
 - ◆ An autonomous system will issue a burn to unload momentum if not otherwise reduced

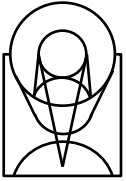


NIRSpec Micro Shutter Array



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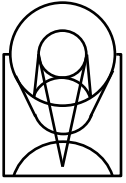


Challenges^(3/5)



■ Nature of the data taken

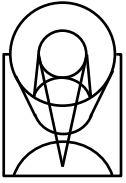
- ◆ As planned design factored in a 2:1 compression of science data for recording and downlink
 - ◆ Realized nature of the data did not lend itself to suitable compression strategies
 - ◆ Initial response was/is to double the number of contacts/day; Was originally 1x 4 hour pass – now 2x
 - ◆ Now considering compressing a series of delta images (readouts) taken against some initial baseline image.
 - Since the delta's only reflect what's changed from read-out to read-out – they are much more suitable to compression approaches



Challenges^(4/5)



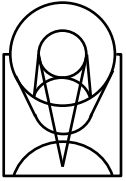
- Dealing with the obsolescence of planned legacy systems
 - ◆ Original plan from 5 years ago cited the usage of then existing systems.
 - ◆ In the 5 years since the plan – several of the planned systems are no longer sustainable for another 5 – 15 years for JWST
 - ◆ OPUS, ETCs, use of the Sunfire/Symetrix, MOSS, use of Sybase, STSDAS/IRAF
 - ◆ Hardware-wise JWST is basing its architecture on new standard configurations
 - ◆ All servers 64-bit Intel based; most using RHEL5; Databases using MS SQL Server



Challenges^(5/5)



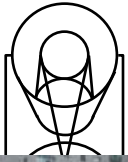
- Dealing with the obsolescence (continued)
 - ◆ Some systems can be re-implemented using a more sustainable approach
 - ◆ MOSS from Fortran to C++; ETCs and necessary STSDAS/IRAF routines being redone in Python
 - ◆ Others require trade studies and evaluations of new approaches
 - ◆ i.e. considering LSST's and NOAO's pipeline systems as replacements for OPUS



Summary



- Use of Legacy systems is beneficial to a new project, but ...
 - ◆ Beware not to freely adopt all aspects of the legacy components
 - ◆ Use only what works for the new mission
 - ◆ Recognize the lifetime potential of the legacy software/hardware
 - ◆ Will it last for the duration of the new mission?
- An iterative development approach works well – but don't ignore the benefit of having up-front baselined requirements and interface specifications
 - ◆ You will always find new functions to incorporate though the development phase – but having a strong base set of requirements up-front will help assure you don't miss anything on the way
 - ◆ Particularly useful if you need to rephase your development efforts
- Whenever possible – Test as you fly
 - ◆ Exercise the system in an operational mode up front (during nominal testing) then you'll be less surprised and better trained when operating for real



JWST Full Scale Model



Seattle



Orlando



Colorado Springs



Montreal



San Diego



Goddard Space Flight Center

ADASS-2009



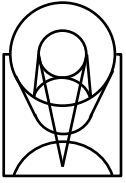
Dublin



5-Oct-09



Munich



JWST Links



- STScI page http://webbtelescope.org/webb_telescope/
- NASA page <http://www.jwst.nasa.gov/index.html>
- ESA page http://www.esa.int/esaSC/120370_index_0_m.html
- Canadian Space Agency:
<http://www.asc-csa.gc.ca/eng/satellites/jwst/default.asp>
- Northrop Grumman:
<http://www.as.northropgrumman.com/products/jwst/index.html>
- U of Arizona page on NIRCAM:
<http://ircamera.as.arizona.edu/nircam/>
- JPL page on MIRI:
<http://www.jpl.nasa.gov/missions/missiondetails.cfm?mission=Webb>